# An Optimal Control Strategy for Multi-Terminal HVDC Systems 

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#### Abstract

An optimal control strategy for the modulation of the DC power of multiterminal HVDC Systems is proposed. The modulated DC power is utilized to enhance the operation of AC systems. The linear programming optimization method along with the feasible direction algorithm was used to obtain the optimum DC current settings. The control strategy has been tested on two hypothetical multi-terminal HVDC systems. The proposed strategy proves to be effective in improving the system response. Another characteristic of this scheme is that improvements of one machine will not be at the expense of other machines. The control scheme will enhance the transient stability if there is a room in the power system to enhance.


Keywords: HVDC systems, Optimal Control, Stability

## 1. Introduction

The potential of utilizing an HVDC system to enhance the performance of an AC system is well known and documented ${ }^{[1-7]}$. Several models have been proposed for the simulation of $\mathrm{AC} / \mathrm{DC}$ systems in transient stability studies. This paper deals with use of the optimization techniques for the control of Multi-Terminals DC (MTDC) systems. Optimization techniques have been applied for the control of HVDC systems. For example, Hamzei proposed a systematic method that is based on the linear programming formulation ${ }^{[8]}$. It is
used to find the DC network solution of a general multi-terminal HVDC system and it's mode of operation. The objective function is to minimize the sum of the cosine function of the control angles. The constraints are the network equations, the control angles and the current orders.

Another area where optimization techniques were applied is correcting the voltages in AC-HVDC systems. The problem was formulated as a linear programming model. The objective function is to minimize the absolute values of the corrections to be made. The constraints are the system voltage and current equations. They also include the power change at the generator buses and the transmission line ratings.

Hamzei and Ong have proposed a model to coordinate the injections of a multi-terminal DC system ${ }^{[9]}$. The objective function was to minimize the active power changes.

This paper reports on the use of the feasible direction algorithm to determine the optimal setting of the HVDC system. Section two introduces the optimization model and the method of solution. The model implementation is described in section three. The simulation results and discussion are presented in section four. Conclusions are drawn in section five.

## 2. The Optimization Model

The feasible direction method is employed to find the best setting of the HVDC side to stabilize the AC system. The method generates an improving feasible direction. It also determines a step size in that direction.

### 2.1 Feasible Direction Algorithm

The class of Feasible Directions Methods solves a non-linear programming problem by moving from a feasible point to an improved feasible point ${ }^{[10-11]}$. That is, given a feasible point $\mathrm{x}_{\mathrm{k}}$, a direction $\mathrm{d}_{\mathrm{k}}$ is determined such that for $\lambda$, oand sufficiently small, the following two (2) properties are true:

1. $\mathrm{x}_{\mathrm{k}}+\lambda \mathrm{d}_{\mathrm{k}}$ is feasible, and
2.The objective function at $\mathrm{x}_{\mathrm{k}}+\lambda \quad \mathrm{d}_{\mathrm{k}}$ has a better value than at $\mathrm{x}_{\mathrm{k}}$. Where $\mathrm{d}_{\mathrm{k}}$ direction along which the variable $\mathrm{x}_{\mathrm{k}}$ improves

The method of feasible direction involves two steps. The first step is the direction generation. In the second step, the step size is determined. These two steps are explained below.

### 2.1.1 Generating Improving Feasible Directions

Given a feasible point $\mathrm{x}_{\mathrm{k}}$, a nonzero vector $\mathrm{d}_{\mathrm{k}}$ is a feasible improving direction if $\nabla f\left(x_{k}\right)^{t} d_{k}>0$, and $d_{k}$ satisfies the set of constraints. The feasible set is determined by a set of linear inequalities and equations given as $\mathrm{A}_{1} \mathrm{~d} \leq 0$, and $\mathrm{Ed}=0$. The improving direction is generated by minimizing $\nabla f\left(x_{k}\right)^{t} d_{k}$ subject to the constraints $\mathrm{A}_{1} \mathrm{~d}_{\mathrm{k}} \leq 0$ and $\mathrm{Ed}_{\mathrm{k}}=0$. This problem is formulated as follows:

Minimize

$$
\begin{equation*}
\nabla f\left(x_{k}\right)^{t} d_{k} \tag{2.1}
\end{equation*}
$$

Subject to

$$
\begin{align*}
& \mathrm{A}_{1} \mathrm{~d}_{\mathrm{k}} \leq 0  \tag{2.2}\\
& \mathrm{Ed}_{\mathrm{k}}=0 \tag{2.3}
\end{align*}
$$

where $A_{1}$ and $E$ are linearized matrices of the original nonlinear problem. They represent the coefficient of the inequality and equality constraints.

### 2.1.2 Step Size Generation

Given that direction generated in the previous step, the next step is to determine the step size in the direction $\mathrm{d}_{\mathrm{k}}$. The bounding interval algorithm is used to determine the range for $\lambda$. The Golden Search then is employed to find the optimal $\lambda_{k}^{*}$. These steps are given below.
Step 1: Find the interval $[\mathrm{a}, \mathrm{b}]$ where $\lambda$ lies, using Interval Bounding algorithm.
Step 2: Use Golden Search Method to obtain $\lambda_{k}^{*}$
Step 3: The new solution will be:

$$
\begin{equation*}
\mathrm{x}_{\mathrm{k}}=\mathrm{x}_{\mathrm{k}}+\lambda_{k}^{*} \mathrm{~d}_{\mathrm{k}} \tag{2.4}
\end{equation*}
$$

where $\lambda_{k}^{*}$ is the optimal value $\lambda$.

The feasible direction method is terminated when $\left\|d_{k}\right\| \leq \varepsilon$.

## 3. Implementation of the Feasible Direction Method

High Voltage DC Transmission systems are known to have the ability to change their transmitted power. This inherent feature of the HVDC systems is utilized for the support of the ac systems in case a transient event occurs. The HVDC system helps to reduce the accelerating power of machines following a fault. The rate and level of the increase in the HVDC power, in response to the ac system needs, is achieved through many techniques. This section presents the development of an optimization model to determine the level of dc power.

In order to build up an appropriate optimization model, the AC-HVDC objective function and constraints are defined. An objective function may be formed to minimize the accelerating power of the machine. The objective function is stated as follows:

$$
\begin{equation*}
\operatorname{Min} \mathrm{Z}=\sum_{i=1}^{N}\left(P_{m i}-P_{e i}\right)^{2} \tag{2.5}
\end{equation*}
$$

N : number of the electrical machines
$\mathrm{P}_{\mathrm{e} i}$ : Electrical output power of the i-th machine
$P_{m i}$ Mechanical input power of the i-th machine
The model constraints are the AC and the HVDC equations. They are solved through Newton-Raphson method. The network constraints have the form given in equation 2.6

$$
\begin{equation*}
\mathrm{R}_{\mathrm{ac} / \mathrm{dc}}=-\mathrm{J}_{\mathrm{ac} / \mathrm{dc}} \Delta \mathrm{X}_{\mathrm{ac} / \mathrm{dc}} \tag{2.6}
\end{equation*}
$$

where:
$\mathrm{R}_{\mathrm{ac} / \mathrm{dc}}$ : a column vector of the ac/dc residuals.
$\mathrm{J}_{\mathrm{ac} / \mathrm{dc}}$ : The Jacobian matrix of the ac/dc network
$\Delta X_{\mathrm{ac} / \mathrm{dc}}$ : a column vector of the changes of the ac/dc variables
The intention is to link the HVDC constraints to the objective function of equation 2.5. Hence equation 2.6 can be rewritten as:

$$
\begin{equation*}
R_{d c}^{\prime}=-\mathrm{J}_{\mathrm{dc}}\left(\Delta \mathrm{X}_{\mathrm{dc}}+\Delta \mathrm{X}_{\mathrm{aux}}\right) \tag{2.7}
\end{equation*}
$$

where:
$R_{d c}^{\prime}:$ column vector of the dc residuals.
$\mathrm{J}_{\mathrm{dc}}$ : a Jacobian matrix of the dc variables.
$\Delta \mathrm{X}_{\mathrm{dc}}$ : a column vector of the changes of the dc variables
$\Delta X_{\text {aux }}$ : a column vectors of the changes in auxiliary variables
Let $\Delta \mathrm{X}_{\text {aux }}$ be equal to $\lambda d$ then equation 2.7 becomes:

$$
\begin{equation*}
\mathrm{R}_{\mathrm{dc}}{ }^{\prime}=-\mathrm{J}_{\mathrm{dc}}\left(\Delta \mathrm{X}_{\mathrm{dc}}+\lambda \mathrm{d}\right) \tag{2.8}
\end{equation*}
$$

The residuals cannot be zero, otherwise a trivial solution is obtained. The residuals are set to small values in order to provide nonzero solution for the direction vector (d).

The vector $\Delta \mathrm{X}_{\mathrm{aux}}$ represents the constraints to the objective function. The Gradient method can be applied to solve this problem. The model becomes as shown in equation 2.9.

$$
\begin{equation*}
\operatorname{Min} \mathrm{Z}=\sum_{i=1}^{N}\left(P_{m i}-P_{e i}\right)^{2} \tag{2.9}
\end{equation*}
$$

subject to:

$$
\begin{equation*}
\mathrm{R}_{\mathrm{dc}}{ }^{\prime}=-\mathrm{J}_{\mathrm{dc}} \Delta \mathrm{X}_{\mathrm{aux}} \tag{2.10}
\end{equation*}
$$

or

$$
\begin{equation*}
\mathrm{R}_{\mathrm{dc}}{ }^{\prime}=-\mathrm{J}_{\mathrm{dc}} \lambda \mathrm{~d} \tag{2.11}
\end{equation*}
$$

Z is the above objective function to be minimized.
The function, $\Delta f(y)$, represents the change in the accelerating power in response to the control variables. It can be numerically calculated and is equal to $-2\left(P_{m i}-P_{e i}\right) * \frac{\partial P_{e i}}{\partial X}$.

Equations 2.7 and 2.9 form the required optimization model. It has a linear form and is solved by using the Golden Search Method. It is worth noting that the objective function is not a direct function of the DC variables. It is, however, a function of the AC variables.

## 4. Results and Discussion

This section describes the results of the implementation of the optimization model when applied to two MTDC power systems. The studies include the assessment of the system performance following a three-phase fault at a major busbar. The evaluation consists of the system response due to:
a. No DC power control i.e. the DC power is not modulated or controlled.
b. Applying a modulating signal to increase the DC power in an arbitrary manner. This is referred to as Control Scheme I.
c. Applying the optimization algorithm without using the Golden Search Technique to determine $\lambda$. This assumes that $\lambda=1.0$. This is referred to as Control Scheme II.
d. Applying the optimization algorithm using Golden Search Technique. This is referred to as Control Scheme III.

An HVDC link is capable of changing the transmitted power, within its rating, in a fast manner. This flexibility in control can vary the transmitted power in accordance to the AC system needs. The power can be modulated in response to any of several signals. These signals give indications of the state of the AC systems. The signals can be:
(a) Speed deviation from synchronous speed, $\Delta \omega$
(b) Rotor slip
(c) Change in generator power
(d) Change in tie-line power
(e) A combination of the above.

The signals are applied through controller of various transfer functions. However, only a simple time-lag controller is used in this paper. The output of the controller is the required change in the DC current. The latter is used to determine the required DC power.

### 4.1 20-Bus System Study

A 20-bus system, whose single line diagram is shown in Fig.1, has been selected for the study. The network belongs to a Middle Eastern utility. It has inherent stability problems. The power system is an AC network. However for the purpose of this study a three terminals MTDC system is proposed. The system data is given in Appendix A1.


Fig. 1. A20-Bus Test System.

The transient behavior of the system will be investigated for two fault locations. The first location is near a generator while the second is close to an inverter terminal.

The response of the rotor angle of machine at ROLL is shown following a 3-phase fault at the generator Bus of BURD. The fault duration is 100 milliseconds. Figure 2 shows the variations of the rotor angle of the machine under the various control schemes. Three signals are applied separately to the controller. These signals are rotor slip, generator power and electric powermechanical difference signals. Control Scheme II uses the linear optimization model derived in section 2 . In this scheme $\lambda$ is assumed to be equal to unity. Control Scheme III uses the Golden Search Method. Golden Search will seek for the optimum $\lambda^{*}$. This, of course, will require freezing the time and iteration to find $\lambda^{*}$. Figure 2 shows the system performance under the three control schemes. Control schemes II \& I tend to overdamp the generators slightly as compared with no control scheme. Control Scheme III improves the behavior slightly.


Fig. 2. Rotor Angle at ROLL due to Fault No. 1 with various Control Strategies.

Figure 3 shows the angular variations of the machine at WADM if the fault location is changed to SWSN which is near the inverter terminal. The fault duration is 0.1 sec . The machine may lose synchronism if the DC power is controlled without the Golden Search technique. The response is much better when the DC power is controlled through the optimization algorithm. This shows that increasing the DC power in arbitrary manner may cause instability in some systems.


Fig. 3. Rotor Angle at WADM due to Fault No. 2 with various Control Strategies.

### 4.2 24-Bus System Study

The second study system is a 24 -bus, 5- machines network. A Threeterminals HVDC system has been proposed. Two faults have been selected, F1 is close inverter no. 1 and F2 is on bus03. Figure 4 shows the single line diagram of the study network. The system data is given in Appendix A2. The rotor angle variation of generator bus04 is shown in Figure 5. The generator is shown to lose stability as a result of the system fault. Both control schemes II and I fail to stabilize the machine. Scheme II is closer to succeed. Only scheme III will stabilize the machine.


Fig. 4. A-24 Bus Test System.


Fig. 5. Rotor Angle of generator at bus04 due to Fault 1with Various Control Schemes.

Figure 6 shows the voltage variations at busbar bus04 following the fault under the various control schemes. The busbar voltage experiences a severe dip to 0.92 p.u. when the control scheme III is adopted. This scheme requires that the generator at bus 04 should supply a large amount of power in order to stabilize the entire system. Consequently, the machine terminal voltage experiences such a dip.


Fig. 6. Voltage Profile at bus04 with various Control Strategies.

Another three-phase fault (F2) was applied on bus03 for 50 ms . the rotor angles response of generator bus04 is shown in Figure 7. The generator loses synchronism due to the fault. Neither of the schemes is able to stabilize the machine. The HVDC link is far from the fault location and thus its impact is negligible.


Fig. 7 Rotor Angle of generator at bus04 due to Fault 2 with various Control Schemes.

## 5. Conclusion

In this paper, three main control schemes have been applied to improve the AC transient stability through HVDC systems. These control schemes were applied to two multimachine systems, on which two three phase faults were applied. The Linear Programming optimization method along with the feasible direction concept proves to be effective in improving the system response. One other characteristic of this scheme over the rest is that an improvement of one machine will not be at the expense of other machines. It shall be noted that the location, duration, type of fault and the power system configuration will affect the contribution of the applied control scheme and the HVDC system. The control scheme will enhance the transient stability if there is room in the power system to enhance. Enhancement cannot be brought from outside the system but from inside by properly and quickly changing the parameters of the MTDC system in the AC power systems.

## Notations

$\mathrm{x}_{\mathrm{k}} \quad$ Feasible point
$\mathrm{d}_{\mathrm{k}} \quad$ Direction along which the variable $\mathrm{x}_{\mathrm{k}}$ improves.
$\lambda \quad$ constant to speedup the improvement in the variable $\mathrm{x}_{\mathrm{k}}$.
$\lambda_{k}^{*} \quad$ The optimal value of $\lambda$
$\mathrm{N}: \quad$ Number of the electrical machines
$\mathrm{P}_{\mathrm{ei}}$ : $\quad$ Electrical output power of the i-th machine
$\mathrm{P}_{\mathrm{mi}} \quad$ Mechanical input power of the i-th machine
$\mathrm{R}_{\mathrm{a} / \mathrm{c} / \mathrm{d}} \quad \mathrm{A}$ column vector of the ac/dc residuals.
$\mathrm{J}_{\text {ac/dc }} \quad$ The Jacobian matrix of the ac/dc network
$\Delta \mathrm{X}_{\text {ac/dc }} \quad$ A column vector of the changes of the ac/dc variables
$R_{d c}^{\prime}: \quad$ Column vector of the dc residuals.
$\mathrm{J}_{\mathrm{dc}} \quad$ A Jacobian matrix of the dc variables.
$\Delta X_{\mathrm{dc}} \quad$ A column vector of the changes of the dc variables
$\Delta X_{\text {aux }} \quad$ A column vector of the changes in auxiliary variables

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## Appendix A1

20-Bus System Data

Busbar Data

| $\begin{array}{\|l} \hline \text { Bus } \\ \text { No. } \\ \hline \end{array}$ | Bus <br> Name | Voltage (PU) | $\begin{aligned} & \text { Angle } \\ & \text { (Deg) } \end{aligned}$ | $\begin{array}{\|l} \hline \text { PG } \\ (M W) \end{array}$ | $\begin{aligned} & \text { QG } \\ & \text { (MVAR) } \end{aligned}$ | $\begin{aligned} & \mathrm{PL} \\ & (\mathrm{MW}) \end{aligned}$ | $\begin{aligned} & \mathrm{QL} \\ & \text { (MVAR) } \end{aligned}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | BURS | 1.00 | 0.00 | 4.00 | 0.00 | 0.00 | 0.00 |
| 2 | BURD | 1.00 | 0.00 | 2.00 | 0.00 | 0.00 | 0.00 |
| 3 | BURM | 1.00 | 0.00 | 0.00 | 0.00 | 13.00 | 10.00 |
| 4 | SWSN | 1.00 | 0.00 | 0.00 | 0.00 | 20.00 | 20.00 |
| 5 | KILM | 1.00 | 0.00 | 0.00 | 8.00 | 6.00 | 0.00 |
| 6 | KUKM | 1.00 | 0.00 | 0.00 | 0.00 | 12.00 | 10.00 |
| 7 | KUKI | 1.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| 8 | KILH | 1.00 | 0.00 | 0.00 | 0.00 | 0.00 | 15.00 |
| 9 | KILI | 1.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| 10 | MERH | 1.00 | 0.00 | 0.00 | 0.00 | 0.00 | 15.00 |
| 11 | ROSH | 1.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| 12 | ROSL | 1.05 | 0.00 | 22.00 | 0.00 | 0.00 | 0.00 |
| 13 | ROLL | 1.04 | 0.00 | 47.40 | 0.00 | 0.00 | 40.00 |
| 14 | HASI | 1.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| 15 | HASL | 1.00 | 0.00 | 0.00 | 0.00 | 6.00 | 4.00 |
| 16 | MERM | 1.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| 17 | WADM | 1.03 | 0.00 | 3.00 | 0.00 | 7.00 | 5.00 |
| 18 | SENM | 1.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| 19 | RABK | 1.00 | 0.00 | 0.00 | 0.00 | 3.00 | 2.00 |
| 20 | SENL | 1.00 | 0.00 | 12.00 | 0.00 | 12.00 | 8.00 |

Machine Parameters

| Bus | Gen. MVA |  | Direct axis transient reactance | Quadrature axis transient reactance | Direct axis synchronous reactance | Quadrature axis <br> synchronous reactance |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| ROSL | 33.50 | 4.50 | 0.28 | 0.40 | 0.77 | 0.40 |
| ROLL | 33.50 | 9.00 | 0.14 | 0.20 | 0.39 | 0.20 |
| SENL | 9.90 | 4.44 | 0.15 | 0.30 | 0.51 | 0.30 |
| BURS | 12.50 | 10.20 | 0.07 | 0.67 | 0.67 | 0.67 |
| BURD | 3.75 | 5.50 | 0.07 | 0.37 | 0.31 | 0.37 |
| WADM | 3.39 | 2.10 | 0.14 | 0.41 | 0.73 | 0.41 |

HVDC Parameters
Terminals

| Rectifier | Inverter | Inverter |
| :--- | :--- | :--- |
| KUKM | SWSN | BURM |

Line Data

|  |  | Resistance <br> FROM | Reactance <br> $($ PU $)$ | SUSC. <br> (PU) | Tap(\%) |
| :--- | :--- | :--- | :--- | :--- | :--- |
| BURS | BURS | 0.00 | 0.384 | 0 | 0 |
| BURD | BURM | 0.00 | 0.213 | 0 | 0 |
| BURM | KILM | 0.21 | 0.358 | 0 | 0 |
| BURM | KUKM | 0.04 | 4.9 | 0 | 0 |
| SWSN | KUKM | 0.02 | 4.9 | 0 | 0 |
| BURM | SWSN | 0.04 | 0.029 | 0 | 0 |
| KUKM | KUKI | 0.00 | 0.116 | 0 | -5.0 |
| KUKI | KILI | 0.0047 | 0.0233 | 0.010 | 0 |
| KILI | KILM | 0.00 | 0.4025 | 0 | -5.0 |
| KILH | KILI | 0.00 | 0.07 | 0 | -2.0 |
| MERH | KILM | 0.02 | 0.14 | 0.2362 | 0 |
| MERH | KILM | 0.02 | 0.14 | 0.2362 | 0 |
| ROSH | MERH | 0.04 | 0.26 | 0.432 | 0 |
| ROSH | MERH | 0.04 | 0.26 | 0.432 | 0 |
| ROSH | ROLL |  | 0.1485 | 0 | 0 |
| ROSH | ROSL |  | 0.297 | 0 | 0 |
| MERH | MERM |  | 0.156 | 0 | 0 |
| MERH | HASI | 0.14 | 0.186 | 0.0175 |  |
| KILI | HASI | 0.307 | 0.407 | 0.038 | 0 |
| HASI | HASL | 0 | 0.636 | 0 | -4.0 |
| MERM | WADM |  | 1.265 | 0 | -2.0 |
| MERM | SENM | 0.2505 | 0.333 | 0.0312 | 0 |
| SENM | RABK | 0.289 | 0.378 | 0.0354 | 0 |
| SENM | SENL | 0 | 1.1 |  | 0 |
| KUKM | KUKM |  | -40 |  |  |
| SWSN | SWSN |  | -40 |  |  |
| BURM | BURM |  | -40 |  |  |


| Specified DC Power |  |  |
| :---: | :---: | :---: |
| PDC | PDC | PD |
| (REC.) | (INV.1) | (INV.2) |
| (MW) | (MW) | (MW) |
| 28.7 | 21 | 7.7 |

Initial Firing Angles (Deg)

| Rectifier | Inverter | Inverter |
| ---: | ---: | ---: |
| 5.00 | 5.00 | 5.00 |

## Appendix $\mathbf{A 2}$

24-Bus System Data

| $\begin{array}{\|l} \hline \text { Bus } \\ \text { No. } \end{array}$ | Bus Name | $\begin{aligned} & \begin{array}{l} \text { Voltage } \\ (\mathrm{PU}) \end{array} \\ & \hline \end{aligned}$ | Angle <br> (Deg) | $\begin{aligned} & \hline \text { PG } \\ & (\mathrm{MW}) \end{aligned}$ | QG (MVAR) | $\begin{aligned} & \begin{array}{l} \mathrm{PL} \\ (\mathrm{MW}) \end{array} \\ & \hline \end{aligned}$ | $\begin{array}{\|l\|} \hline \text { QL } \\ \text { (MVAR) } \end{array}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | bus01 | 1.00 | 0.00 | 78.00 | 40.00 | 0.00 | 0.00 |
| 2 | bus02 | 1.00 | 0.00 | 100.00 | 20.00 | 0.00 | 0.00 |
| 3 | bus03 | 0.99 | 0.00 | 25.00 | 8.9 | 0.00 | 0.0 |
| 4 | bus04 | 0.96 | 0.00 | 76.00 | 42.00 | 0.00 | 0.0 |
| 5 | bus05 | 1.05 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| 6 | bus06 | 1.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| 7 | bus07 | 1.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| 8 | bus08 | 1.00 | 0.00 | 0.00 | 0.00 | 104.99 | 52.0 |
| 9 | bus09 | 1.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| 10 | bus10 | 1.00 | 0.00 | 0.00 | 0.00 | 18.00 | 14.00 |
| 11 | bus11 | 1.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| 12 | bus12 | 1.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| 13 | bus13 | 1.00 | 0.00 | 0.00 | 0.00 | 61.00 | 32.00 |
| 14 | bus14 | 1.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.0 |
| 15 | bus15 | 1.00 | 0.00 | 0.00 | 0.00 | 6.00 | 4.00 |
| 16 | bus16 | 1.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| 17 | bus17 | 1.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| 18 | bus18 | 1.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| 19 | bus19 | 1.00 | 0.00 | 0.00 | 0.00 | 51.00 | 10.00 |
| 20 | bus20 | 1.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.0 |
| 21 | bus21 | $1.00$ | 0.00 | 0.00 | 0.00 | 12.00 | 6.00 |
| 22 | bus22 | 1.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| 23 | bus23 | 1.00 | 0.00 | 0.00 | 0.00 | 28.00 | 10.00 |
| 24 | bus24 | 1.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |

Machine Parameters

| Bus | Gen. <br> MVA | Inertia constant H (sec) | Direct axis transient <br> reactance | Quadrature axis transient reactance | Direct axis <br> synchronous <br> reactance | Quadrature axis <br> synchronous <br> reactance |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| bus01 | 100 | 5.08 | 0.2005 | 0.2005 | 1.305 | 0.85 |
| bus02 | 100 | 1.892 | 0.3235 | 0.3235 | 0.9712 | 0.6325 |
| bus03 | 100 | 1.690 | 0.523 | 0.523 | 4.72 | 4.70 |
| bus04 | 100 | 4.400 | 0.200 | 0.200 | 1.970 | 1.953 |
| bus05 | 100 | 0.248 | 1.55 | 1.55 | 5.8 | 3.75 |

## HVDC Parameters

Terminals

| Rectifier | Inverter | Inverter |
| :--- | :--- | :--- |
| bus13 | bus12 | bus16 |

Specified DC Power

| PDC | PDC | PDC |
| ---: | :--- | :--- |
| (REC. $)$ | (INV.1) | (INV.2) |
| $(\mathrm{MW})$ | $(\mathrm{MW})$ | $(\mathrm{MW})$ |
| 18 | 9 | 9 |

Initial Firing Angles (Deg)

| Rectifier | Inverter | Inverter |
| ---: | ---: | ---: |
| 5.00 | 15.00 | 15.00 |


| Line Data |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| FROM | TO | Resistance <br> (PU). | Reactance <br> (PU) | SUSC <br> (PU) | Tap |
| bus06 | bus02 | 0.00 | 0.09 | 0 | 10.0 |
| bus06 | bus07 | 0.00476 | 0.0199 | 0.01660 |  |
| bus07 | bus01 | 0.00 | 0.1125 | 0 | 12.5 |
| bus08 | bus24 | 0.0719 | 0.3178 | 0.061 |  |
| bus08 | bus15 | 0.0516 | 0.1062 | 0.0203 |  |
| bus08 | bus09 | 0.00 | 0.2438 | 0 | -5.0 |
| bus09 | bus05 | 0.00 | 0.106 | 0 |  |
| bus15 | bus16 | 0.00 | 0.333 |  | -1.0 |
| bus09 | bus10 | 0.00 | 0.00629 | 0 |  |
| bus08 | bus11 | 0.0516 | 0.1062 | 0.0203 |  |
| bus11 | bus12 | 0.00 | 0.333 |  | -1.0 |
| bus12 | bus14 | 0.00 | 0.10 | 0.0 |  |
| bus13 | bus04 | 0.00 | 0.106 | 0.00 | 15.0 |
| bus18 | bus19 | 0.00 | 0.20 | 0 | 4.0 |
| bus18 | bus20 | 0.0208 | 0.0807 | 0.0754 |  |
| bus20 | bus21 | 0.00 | 0.333 |  | -1.0 |
| bus20 | bus22 | 0.0246 | 0.0967 | 0.0777 |  |
| bus22 | bus23 | 0.00 | 0.1667 |  | 2.0 |
| bus07 | bus08 | 0.0361 | 0.145 | 0.13 |  |
| bus07 | bus18 | 0.0148 | 0.054 | 0.0536 |  |
| bus14 | bus14 |  | -20 |  |  |
| bus17 | bus17 |  | -20 |  |  |
| bus02 | bus02 |  | -40 |  |  |

# لمستراليجية تحكم مثلل لأظمة التيار المستمر <br> <br> متعدد الأطرف 

 <br> <br> متعدد الأطرف}

إبراهيم الأمين
جلمعة المك فهـ للبترول والمعان الظهرلن 31261

المستخلص : تعتبر الاسنقرارية العاب ـرة جزءًا هلمًا ف في تص ميم
 المستمر متعدد الطٔطرف لتهسين أداء الاسققرارية العابرة لل ـرعة تجاوبها في تبيل الطالة الكهربائية للأظمة ذا لت التيار المتردد كلأحد المزايا لها. وتد مَ عمل نموذج رياضي من لجُل تمثل أظمة التيار المستمر متعدد الأطرف. وقد مَ وضع جمبع المعادلات الفاض ليا للظظل وآليت التحكم في البحث علىشكل معادلات جبرية ومن مِ حل

 الاعتدالهي للوصول إله حل أفضل كطرقة لأخرى مقترحة . ولجريت أيضًا درلسة لنوعين من آليت التحكم القلقيية تعمل علا مى لإث الخارة

 المولدت حيث أنل عليها ظلم التي ـار المس تمر ذي الأط ـرف الثلاثة. وفي النهاية لجريت درلمة مقارنة بين الط ـرق الـو الم الـكورة
 الخطية مع ظرية البهث الاعتدالي أعطت أفضل النتائج لتدب ــن الين الالسقرارية العابرة لأظمة التيار المتردد.

